

SOME FEATURES OF TEMPERATURE FIELDS IN THE COMBUSTION CHAMBER OF A GAS TURBINE ENGINE

Ya. V. Tropin^{1,2} and S. A. Rashkovskiy²

¹A. Lyulka Experimental Design Bureau, Branch of the United Engine Corporation — Ufa Engine Industrial Association, 13 Kasatkina Str., Moscow 129301, Russian Federation

²A. Yu. Ishlinsky Institute for Problems in Mechanics of the Russian Academy of Sciences, 101 Vernadskiy Av., Moscow 119526, Russian Federation

Abstract: Based on the analysis of experimental data, periodic specific features of temperature fields at the outlet of the flame tube of the combustion chamber of a gas turbine engine have been revealed. It is shown that, contrary to expectations, the temperature fields in the central belt of the outlet cross section of the flame tube do not reflect the individuality of each burner device: the number of maxima and minima of the temperature field is twice less than the number of combustion chamber burners. Thus, it was found for the first time that at the outlet of the flame tube, there is a pairwise integration of temperature fields created by neighboring burners. It was found that the pairwise integration of the temperature fields of neighboring burners is not random: in different experiments, the temperature fields of the same burners are always combined. This indicates that the reason for such integration is some design feature of the combustion chamber. Based on the analysis performed, it is concluded that the cause of the pairwise integration of temperature fields are the racks located at the combustion chamber inlet but the nature of this effect remains unclear.

Keywords: gas turbine engine; combustion chamber; flame tube; temperature fields; combustion

DOI: 10.30826/CE24170204

EDN: VBOSUL

Figure Captions

Figure 1 Schematic diagram of the installation for testing the annular combustion chamber [15]: 1–10 — thermocouples; and arrow in (b) (view from the turbine side) — direction of rotation of the turret with thermocouple combs. The racks in the combustion chamber inlet diffuser are not shown in (a)

Figure 2 Schematic (sweep) of the combustion chamber inlet part: 1 — rack; 2 — flame tube mountings; and 3 — burner devices with swirl vanes

Figure 3 Average temperature fields from thermocouple data of the central (a), inner (b), and outer belts (c)

Figure 4 Fourier series coefficients for different modes of the central (a), inner (b), and outer belt temperature fields (c)

Acknowledgments

The work was supported by the Ministry of Science and Higher Education within the framework of the Russian State Assignment under contract No. 124012500440-9.

References

1. Inozemtsev, A. A., M. A. Nikhamkin, and V. L. Sandratskiy. 2008. *Osnovy konstruirovaniya aviatsionnykh dvigateley i energeticheskikh ustanovok* [Fundamentals of design of aircraft engines and power plants]. Moscow: Mashinostroenie. Vol. 2. 367 p.
2. Lefebvre, A. H. 1983. *Gas turbine combustion*. West Lafayette, IN: Purdue University. 531 p.
3. Mingazov, B. G. 2006. *Kamery sgoraniya gazoturbinnyykh dvigateley. Konstruktsiya, modelirovanie protsessov i raschet* [Combustion chambers of gas turbine engines. Design, process modeling, and calculation]. Kazan: Publishing House of Kazan State Technical University. 220 p.
4. Krylov, B. A., I. I. Onishchik, and A. A. Yun. 2009. Modelirovanie protsessov teplo- i massoobmena v model'nykh kamerakh sgoraniya [Modeling of heat and mass transfer processes in model combustion chambers]. *Aerospace MAIJ*. 16(1):3.
5. Kostyuk, V. Ye., Ye. I. Kirilash, A. V. Stasyuk, et al. 2012. Chislennoe modelirovanie prostranstvennogo raspredeleniya temperatury gaza v polnorazmernoy kamere sgoraniya gazoturbinnogo dvigatelya i ee odnogorelochnom sektore [Numerical simulation of gas temperature spatial distribution in a full-size gas turbine combustor and its one-burner sector]. *Vestnik Samarskogo gosudarstvennogo aerokosmicheskogo universiteta im. akademika S. P. Koroleva (natsional'nogo issledovatel'skogo universiteta)* [Vestnik of the Samara State Aerospace University] 2(33):65–73.

6. Vafin, I. I., and B. G. Mingazov. 2013. Modelirovanie protsessa smesheniya v kamerakh sgoraniya GTD [Modeling of mixing process in combustion chambers of gas turbine engines]. *7th Conference (International) "Problems and Prospects of Development of Aviation, Ground Transportation, and Power Engineering" Proceedings*. 395–398.
7. Sabirzyanov, A. N., V. B. Yavkin, Yu. B. Aleksandrov, *et al.* 2013. Modelirovanie emissionnykh kharakteristik kamer sgoraniya GTD [Modeling of emission characteristics of combustion chambers of gas turbine engine]. *7th Conference (International) "Problems and Prospects of Development of Aviation, Ground Transportation, and Power Engineering" Proceedings*. 1:355.
8. Sipatov, A. M., K. A. Shilov, A. D. Nugumanov, and T. V. Abramchuk. 2016. Chislennaya dovodka poley temperatury gazov na vykhode iz kamery sgoraniya gazoturbinnoy ustanovki [Numerical simulation for fine-tuning the gas temperature fields at the outlet of the gas turbine combustion chamber]. *PNRPAerospace Engineering Bulletin* 3(46):40–55. doi: 10.15593/2224-9982/2016.46.02.
9. Mark, C. P., and A. Selwyn. 2016. Design and analysis of annular combustion chamber of a low bypass turbofan engine in a jet trainer aircraft. *Propulsion Power Research* 5(2):97–107.
10. Agulnik, A. B., I. I. Onishchik, and A. D. Yarmash. 2017. Protsess smesheniya i neravnomernost' polya temperatur gaza na vykhode iz kamery sgoraniya GTD [Mixing process and nonuniformity of the gas temperature field at the outlet of the combustion chamber of a GTE]. *Nasosy. Turbiny. Sistemy* [Pumps. Turbines. Systems] 2(23):30–38.
11. Tao, W., J. Wang, R. Mao, X. Wang, C. Zhang, and Y. Lin. 2019. Generation and migration of hot streaks within an LPP combustor. *Turbo Expo: Turbomachinery Technical Conference and Exposition*. Phoenix, AZ: American Society of Mechanical Engineers. 4A:GT2019-90601. 9 p. doi: 10.1115/GT2019-90601.
12. Bertini, D., L. Mazzei, A. Andreini, and B. Facchini. 2019. Multiphysics numerical investigation of an aeronautical lean burn combustor. *Turbo Expo: Turbomachinery Technical Conference and Exposition*. Phoenix, AZ: American Society of Mechanical Engineers. 5B:GT2019-91437. 15 p. doi: 10.1115/GT2019-91437.
13. Vignat, G., D. Durox, T. Schuller, and S. Can-
del. 2020. Combustion dynamics of annular systems. *Combust. Sci. Technol.* 192(7):1358–1388. doi: 10.1080/00102202.2020.1734583.
14. Tenango-Pirín, O., S. Espinosa, J. C. García, and J. A. Rodríguez. 2024. Analysis of gas turbine combustor exhaust emissions: Effects of transient inlet air pressure. *Energy Technol. — Ger.* 12(3):2301093. doi: 10.1002/ente.202301093.
15. Leontev, M. K. 2008. *Atlas detaley i uzlov dvukhkонтorno-goturboreaktivnogo dvigatelya AL-31F* [Atlas of parts and assemblies of the AL-31F twin-circuit turbojet engine]. Moscow: MAI Publishing House. 20 p.
16. Vyunov, S. A., and Yu. I. Gusev. 1989. *Konstruktsiya i proektirovanie aviatsionnykh gazoturbinnnykh dvigateley* [Construction and design of aviation gas turbine engines]. Moscow: Mashinostroenie. 228–258.
17. OST 1.01134-86. 1986. Kamery sgoraniya osnovnykh gazoturbinnnykh dvigateley. Metody opredeleniya polya temperatury gaza na vykhode iz osnovnoy KS pri avtonomnykh ispytaniyakh [Combustion chambers of main gas turbine engines. Methods for determining the gas temperature field at the main compressor station outlet during autonomous tests]. Moscow. 16 p.
18. Marchukov, E. Yu., I. I. Onishchik, B. B. Rutovskiy, *et al.* 2004. *Ispytaniya i obespechenie nadezhnosti aviatsionnykh dvigateley i energeticheskikh ustanovok* [Testing and ensuring the reliability of aircraft engines and power plants]. Moscow: MAI Publishing House. 334 p.

Received February 26, 2024

Contributors

Tropin Yaroslav V. (b. 1996) — design engineer, A. Lyulka Experimental Design Bureau, Branch of the United Engine Corporation — Ufa Engine Industrial Association; PhD student, A. Yu. Ishlinsky Institute for Problems in Mechanics of the Russian Academy of Sciences, 101 Vernadskiy Av., Moscow 119526, Russian Federation; tropin.ya@yandex.ru

Rashkovskiy Sergey A. (b. 1957) — Doctor of Science in physics and mathematics, chief research scientist, A. Yu. Ishlinsky Institute for Problems in Mechanics of the Russian Academy of Sciences, 101 Vernadskiy Av., Moscow 119526, Russian Federation; rash@ipmnet.ru