

## LOCAL VELOCITIES OF HOT-SPOT COMBUSTION FRONT IN HMX

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**Abstract:** The mechanism of HMX combustion at pressures of 0.05–5.0 MPa is investigated. It is shown that HMX burns in the focal (hot-spot) mode. In the article “HMX combustion mechanism” by V. N. Marshakov, V. G. Krupkin, and S. A. Rashkovsky (2020. Russ. J. Phys. Chem. B 14(6):934–939. doi: 10.1134/S1990793120060111), the scale of inhomogeneity of the combustion surface — the characteristic size of hot spots is determined. The dependence of the size of the hot spots on the average burning rate of the sample is obtained. In the present article, the temperature distributions in the combustion wave obtained using thermocouples are analyzed. The local burning rates are obtained from the analysis of temperature distributions in the condensed phase (close to the Mikchelson distribution). It is shown that the scatter of the burning rate values is explained by the registration of the velocity at different points of the transverse wave front. The values of the local burning rates exceeding the average burning rate are caused by the elevated initial temperature of the sample ahead of the flame front and the smaller values are explained by the curvature of the flame front and by the buckling mode of combustion.

**Keywords:** HMX; multidimensional combustion front; mechanism of hot-spot combustion; transverse waves; local burning rate

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## Figure Captions

**Figure 1** Dependence of the burning rate on the pressure: 1 — our data, 2 — [7, 9, 11], 3.1–3.4 — [5], 4 — [12], 5 — [4], and 6 — [1]

**Figure 2** Temperature profiles of the HMX combustion wave at a pressure of 0.05–4 MPa (1–3): (a) three Π-shaped thermocouples; and (b) the same profiles in the condensed phase in semilogarithmic coordinates with the release of straight-line segments to determine the local burning rate (1 —  $\ln T_1$ , 0,026–0,04 cm/s; 2 —  $\ln T_2$ , 0,023 cm/s; and 3 —  $\ln T_3$ , 0,024–0,033–0,022 cm/s)

**Figure 3** Temperature profiles of the HMX combustion wave: (a)  $p = 0.075$  MPa, three Π-shaped thermocouples; and (b)  $p = 0.1$  MPa, Λ-shaped thermocouple

**Figure 4** Temperature profiles of the HMX combustion wave at  $p = 0.5$  MPa: (a) four Π-shaped thermocouples, and (b) the same profiles in the condensed phase in semilogarithmic coordinates

**Figure 5** Temperature profiles of the HMX combustion wave at  $p = 1.1$  MPa: (a) three Π-shaped thermocouples; and (b)  $p = 2.0$  MPa, four Λ-shaped thermocouples

**Figure 6** Temperature profiles of the HMX combustion wave at  $p = 4.0$  MPa: (a) three Π-shaped thermocouples; and (b) the same profiles in the condensed phase in semilogarithmic coordinates,  $T_s = 490^\circ\text{C}$  (1 —  $\ln T_1$ , 0,89–0,64 cm/s; 2 —  $\ln T_2$ , 0,89–0,45 cm/s; and 3 —  $\ln T_3$ , 0,89–0,36 cm/s)

**Figure 7** The scatter of the values of the local burning rate at different pressures: 1 — 0.05 and 0.075 MPa; 2 — 0.1 MPa; 3 — 0.1 MPa [12]; 4 — 0.5 MPa; 5 — 1.0 [1] and 1.1 MPa; 6 — 2.0 and 2.3 MPa; 7 — 4.0 MPa [1]; vertical line segments — the spread of  $U_n$  values in other experiments;  $A - U_{av} = 2.39p^{0.81}$ ;  $B - U = U_{av}/1.65$ ; and  $C - U = U_{av} \cdot 1.65$

**Figure 8** Burning rate vs. pressure and initial temperature: 1 — our data,  $T_0 = 20^\circ\text{C}$ ; 2 —  $T_0 = 20^\circ\text{C}$  [9]; 3 —  $T_0 = 100^\circ\text{C}$  [1]; 4 —  $T_0 = 100^\circ\text{C}$  [9];  $A - U_{av} = 2.39p^{0.81}$ ; and  $B$  — approximation according to data 3 and 4

## Table Caption

The value of the local burning rate  $U_n$  (mm/s) and pressure  $p$  (MPa)

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## References

1. Puchkov, V. M. 1978. Struktura zon goreniya TRT v stacionarnykh rezhimakh i pri pogasanii [The structure of the combustion zones of SRP in the stationary modes and with the extinction]. Moscow: IChF AN SSSR. PhD Diss. 211 p.
2. Price, C. F., T. L. Boggs, and R. L. Derr. 1979. The steady-state combustion behavior of ammonium perchlorate and HMX. AIAA Paper No. 79-0164. 14 p.
3. Shchepelin, Yu. A., and S. B. Umblia. 1983. Experimental'noe issledovanie mekhanizma goreniya oktogenika [Experimental study of the mechanism of octogen combustion]. *Voprosy vosplamneniya i goreniya raketnykh topliv* [Problems of ignition and combustion of rocket fuels]. Tomsk: TGU. 105–111.
4. Korobeinichev, O. P., L. V. Kuibida, and V. Zh. Hadirbaev. 1984. Investigation of the chemical structure of the HMX flame. *Combust. Explos. Shock Waves* 20(3):282–285.
5. Glazkova, A. P., G. T. Aphanasyev, and S. I. Postnov. 1991. Deflagration and high-temperature ignition of RDX and HMX. *2nd Beijing Symposium (International) on Pyrotechnics and Explosive combined with 17th Pyrotechnics Seminar (International)*. Beijing, People's Republic of China. 1:636.
6. Kubota, N. 1992. Flame structure of modern solid propellants. *Nonsteady burning and combustion stability of solid propellants*. Eds. L. De Luca, E. W. Price, and M. Summerfield. Progress in astronautics and aeronautics ser. Washington, D.C.: AIAA. 143:233–259.
7. Zenin, A. A. 1995. HMX and RDX. Combustion mechanism and influence on modern double-base propellant combustion. *J. Propul. Power* 11(4):752–758.
8. Simonenko, V. N., A. B. Kiskin, V. E. Zarko, and A. G. Svit. 1997. Specific features of nitramine combustion at atmospheric pressure. *Combust. Explos. Shock Waves* 33(6):685–687.
9. Zenin, A. A., V. M. Puchkov, and S. V. Finjakov. 1998. Characteristics of HMX combustion waves at various pressures and initial temperatures. *Combust. Explos. Shock Waves* 34(2):170–176.
10. Atwood, A. L., T. L. Boggs, P. O. Curran, and D. M. Hanson-Parr. 1999. Burning rate of solid propellant ingredients, part 1: Pressure and initial temperature effects. *J. Propul. Power* 15(6):740–742.
11. Zenin, A. A., and S. V. Finjakov. 2006. Characteristics of octogen and hexogen combustion: A comparison. *Prog. 37th Annual Conference (International) of ICT*. Karlsruhe. Paper 118. 18 p.
12. Sinditskii, V. P., V. Yu. Egorshev, M. V. Berezin, and V. V. Serushkin. 2009. Mechanism of HMX combustion in a wide range of pressures. *Combust. Explos. Shock Waves* 45(4):461–477.
13. Marshakov, V. N., V. G. Krupkin, and S. A. Rashkovsky. 2020. HMX Combustion Mechanism. *Russ. J. Phys. Chem. B* 14(6):934–939. doi: 10.1134/S1990793120060111.
14. Zel'dovich, Ya. B. 1942. K teorii goreniya porokhov i vzryvchatykh veshchestv [On the theory of combustion of gunpowder and explosives]. *JETF* 12(11–12):498–524.
15. Novozhilov, B. V. 1973. *Nestatsionarnoe gorenie tverdykh raketnykh topliv* [Nonstationary combustion of solid rocket propellants]. Moscow: Nauka. 176 p.
16. Kondrikov, B. N., and B. V. Novozhilov. 1974. Critical combustion diameter of condensed substances. *Combust. Explos. Shock Waves* 10(5):580–587.
17. Romanov, O. Ya. 2007. Critical combustion diameter. *Combust. Explos. Shock Waves* 43(1):25–33.
18. Rashkovskiy, S. A. 2011. Effect of the curvature of the burning surface of condensed energetic materials on the burning rate. *Combust. Explos. Shock Waves* 47(6):687–696. doi: 10.1134/s0010508211060104.

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