EXPERIMENTAL STUDIES OF SMALL SAMPLES BENCH ROCKET ENGINE WITH A CONTINUOUSLY-DETONATION COMBUSTORS

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Abstract: A test bench for fire tests of demonstration samples of rocket engines of a new type with continuous-detonation combustion has been created. Demonstration tests with small-scale samples of rocket engines with annular combustors 50 and 100 mm in outer diameter with an annular gap of 5 mm, operating on hydrogen–oxygen mixture, proved experimentally for the first time that a thermodynamic Zel'dovich cycle with continuous detonation combustion of hydrogen–oxygen mixture is, respectively, 6%–7% and 7%–8% more efficient than the thermodynamic cycle with continuous constant-pressure combustion of the same mixture under similar conditions. In a series of demonstration tests with the 100-millimeter-diameter combustor using methane–oxygen mixture with an overall equivalence ratio from 0.95 to 1.4, only modes with continuous-detonation combustion were obtained whereas the mode with continuous combustion was not observed. In view of the fact that the pressure in the combustion chambers was relatively low so far (less than 3 atm) and the design of the combustion chambers and nozzles was unoptimized, the values of specific impulse were relatively low (up to 160 s).

Keywords: continuous-detonation combustion; test bench; rocket engine; hydrogen– oxygen; methane–oxygen; Zel'dovich cycle; energy efficiency

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